

### Acknowledgment

This work was supported by NASA MSFC Grant NAS8-38425.

### References

- <sup>1</sup>Krzczkowski, S. A., "Measurement of Liquid Droplet Disintegration Mechanisms," *International Journal of Multiphase Flow*, Vol. 6, No. 1, 1980, pp. 227-239.
- <sup>2</sup>Gonor, A. L., and Zolotova, N. V., "Spreading and Breakup of a Drop in a Gas Stream," *Acta Astronautica*, Vol. II, No. 2, 1984, pp. 137-142.
- <sup>3</sup>Taylor, G. I., "The Shape and Acceleration of a Drop in a High Speed Air Stream," *The Scientific Papers of G. I. Taylor*, edited by G. K. Batchelor, Vol. III, Univ. Press, Cambridge, UK, 1963.
- <sup>4</sup>O'Rourke, P. J., and Amsden, A. A., "The Tab Method for Numerical Calculation of Spray Droplet Breakup," Society of Automotive Engineers Paper 87-2089, 1987.
- <sup>5</sup>Clark, M. M., "Drop Breakup in a Turbulent Flow-I, Conceptual and Modeling Considerations," *Chemical Engineering Science*, Vol. 43, No. 3, 1988, pp. 671-679.

## Liftoff Characteristics of Methane Jet Diffusion Flames

J. P. Seaba\* and L.-D. Chen†

University of Iowa, Iowa City, Iowa 52242  
and

W. M. Roquemore‡  
Wright Laboratory,

Wright-Patterson Air Force Base, Ohio 45433

### Nomenclature

- $d$  = inner diameter of burner exit  
 $R$  = inside radius of burner exit  
 $r$  = radial position from center of burner exit  
 $Uc$  = average centerline velocity  
 $Um$  = average axial velocity  
 $Urms$  = root mean square velocity  
 $X_i$  = mole fraction of fuel with inert gas  $i$   
 $x$  = position normal to burner exit

### Introduction

THE mechanisms responsible for the liftoff from the burner rim and stabilization of jet flames are not clearly understood.<sup>1</sup> Early works<sup>2,3</sup> studied the stability of jet flames in open air and identified four different regimes concerning flame stability. Of the four stability regimes 1) liftoff, 2) blowoff, 3) lifted, and 4) blowout, only the liftoff process will be studied in this Note. The liftoff condition is referred to the instant when the flame detaches from the burner rim in a discontinuous manner. After the flame detaches from the burner rim, it may stabilize at a downstream location (i.e., lifted flame) or it may result in flameoff conditions (or the blowoff condition).

More recent works<sup>4-7</sup> have determined important parameters such as jet exit lip thickness, annular velocity, diluents, etc., which significantly effect the liftoff process. Current data are inconclusive to determine the mechanism(s) responsible for the liftoff process. The present study adds new information to the data base and reveals interesting comparisons to previous liftoff data.

### Experimental Consideration

The experiments were conducted with a small vertical combustion tunnel.<sup>8</sup> The fuel and inert gas flows are regulated by rotameters. The burner is positioned vertically upward, at the center of the coflowing annular air that has a diameter of 250 mm. The burner tip extends 25-mm above the annulus exit plane. A stainless steel honeycomb mesh is placed in the annulus perpendicular to the axial direction, upstream of the burner exit. The burner assembly is situated under a forced-draft exhaust hood (1.22 × 1.22 m) with window screens surrounding the combustor. The exhaust system is equipped with a blower rated at 0.944 m<sup>3</sup>/s. The dry annulus air (dew point at -40°C) is metered by a mass flow meter. The inert gas is added only to the fuel flow. A pressure gauge downstream of the rotameters is used to monitor the pressure drop between the rotameters and the burner exit. The pressure drop for the conditions reported here is less than 0.69 kPa, which has a negligible back-pressure effect relative to the rotameter reading. The flow meters are specified with an accuracy of ±5% of the full-scale reading. Liftoff conditions were recorded with and without the exhaust hood operating; no difference in the liftoff velocities was observed.

The burner configuration is a tapered nozzle. The burner exit i.d. is 5 mm. The burner is made of a long stainless steel tube, having an o.d. of 25.4 mm, and a reducing section near the burner tip. The tapered nozzle has a gradually tapered section, at a rate of 1-20 radius-to-length ratio, over a length of 100 mm, forming a 2.9-deg angle to the vertical plane. The tapered burner has a sharp lip, about 0.3 mm in thickness, and yields a flat velocity profile at the burner exit as shown in Fig. 1.

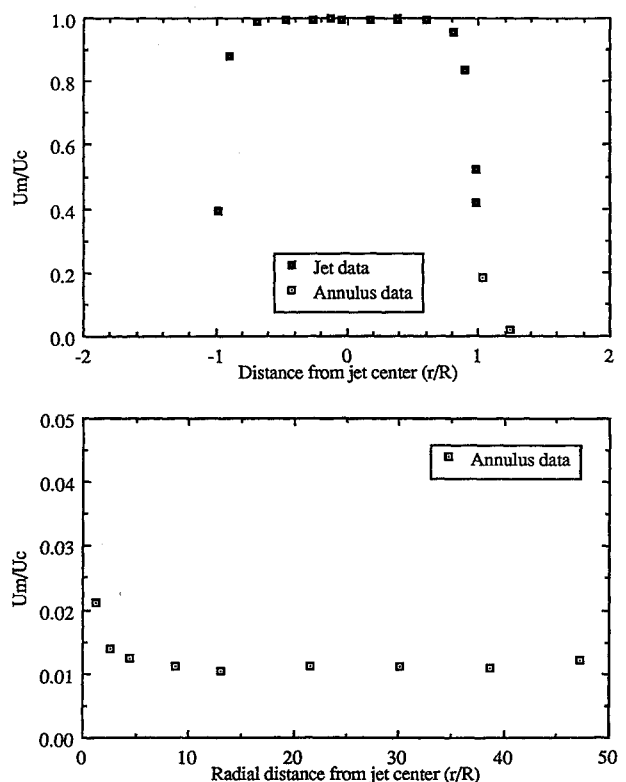


Fig. 1 Average velocity profile using argon jet at condition B,  $x/d = 0.4$ .

Received July 13, 1991; revision received Feb. 25, 1993; accepted for publication Feb. 28, 1993. Copyright © 1993 by the American Institute of Aeronautics and Astronautics, Inc. All rights reserved.

\*Graduate Research Assistant, Department of Mechanical Engineering; currently Assistant Professor, Department of Mechanical and Aerospace Engineering, University of Missouri—Columbia, Columbia, MO 65211. Member AIAA.

†Professor, Department of Mechanical Engineering. Member AIAA.

‡Aero Propulsion and Power Directorate. Member AIAA.

Liftoff conditions of methane ( $\text{CH}_4$ ) diluted with three inert gases 1) nitrogen ( $\text{N}_2$ ), 2) helium ( $\text{He}$ ), and 3) argon ( $\text{Ar}$ ), were experimentally determined. The purity of the gases is better than 97% for methane (technical grade), and 99.9% for the diluents. The fuel mole fraction at the jet exit varies from 0.3 to 1. Three different annular airflow rates were used, conditions are identified as conditions A, B, and C in the discussion herein (average velocities of 0.07, 0.15, and 0.30 m/s, respectively). These velocities were based on the mass flow readings with an accuracy of  $\pm 0.005$  m/s. The liftoff velocities were determined by setting a fixed annulus airflow rate, then increasing the jet velocity until the flame lifted from the nozzle exit and stabilized at a downstream location. The fuel-diluent liftoff velocities were obtained by setting the fuel flow rate (less than the liftoff velocity) and increasing the diluent flow rate until liftoff. This procedure was repeated several times at each condition. The reproducibility of the liftoff velocity is within  $\pm 2\%$  of the results reported in this Note. The fuel jet velocity reported here is also based on the rotameter readings.

A single-channel laser Doppler velocimetry (LDV) system was used to quantify the exit condition of the combustor. The LDV system was operated at 514.5 nm in the forward scattering mode. A burst counter was used to process the signal. A microcomputer was used for data acquisition and to analyze the data. Details of the LDV system are given in Ref. 1. The LDV system collected a minimum of 3000 velocity measurements at each location as shown in Figs. 1 and 2. A conditional seeding technique was used, where either the jet or annular flow was seeded with  $1\text{-}\mu\text{-diam}$   $\text{Al}_2\text{O}_3$  particles. All locations displayed a Gaussian profile. A time-weighted averaging technique was used to determine the average and root-mean-squared velocities.<sup>1</sup> The average velocity measurements have a maximum uncertainty of  $\pm 3\%$ .<sup>1</sup>

## Results and Discussion

The fuel jet and annular air have relatively uniform axial velocities at the combustor exit. As an example, the axial velocity of an argon jet at  $x/d = 0.4$  (i.e., at near injector exit) is shown in Fig. 1. The Reynolds number of the argon jet is 4700 and has the same momentum flux at the burner exit as the methane jet at the near lift-off condition with annular air set at condition B (0.15 m/s). The argon jet has a flat velocity profile extending from  $r/R = 0$  to  $r/R = 0.8$ , and a sharp decrease beyond  $r/R = 0.8$ . The velocity decreases to 0.1 of its centerline value at  $r/R = 1.2$ . The annular air also has a relatively uniform velocity profile which is shown in Fig. 1. The turbulence level ( $U_{rms}/U_c$ ) is approximately 1% at the jet center ( $0 < r/R < 0.8$ ), 0.2% in the outer annulus ( $r/R > 1.2$ ), and obtains a maximum value of 14% in the jet shear layer ( $0.8 < r/R < 1.2$ ). The turbulence level profile is shown in Fig. 2.

Stability curves of methane jet diffusion flames diluted by Ar,  $\text{N}_2$ , or He are shown in Fig. 3. The stability curve is defined as the fuel jet lift-off velocity vs initial fuel mole fraction. The annular air velocity is a parameter in the construction of the stability curve. Three annular air velocities are considered, i.e., conditions A, B, and C.

The stability curves show that the jet lift-off velocity decreases as fuel is diluted by the inert gases, i.e.,  $\text{N}_2$ , Ar, or He. The stability curves also show that the lift-off velocity decreases when the annular air velocity is increased. The annular-air effect is in agreement with Takahashi<sup>7</sup> in that the fuel jet lift-off velocity decreased when the coflowing air velocity is increased. It is also clear from the stability curves presented in Fig. 3 that the fuel jet lift-off velocity decreases when the dilution concentration is increased. At condition A, the fuel jet lift-off velocity decreases from 31.4 m/s for pure methane (i.e.,  $X_F = 1$ ) to 0.94 m/s for argon dilution at  $X_{Ar} = 0.29$ ; it decreases to 0.74 m/s for nitrogen dilution at  $X_{N_2} = 0.4$ ; and it decreases to 0.66 m/s for helium dilution at  $X_{He}$

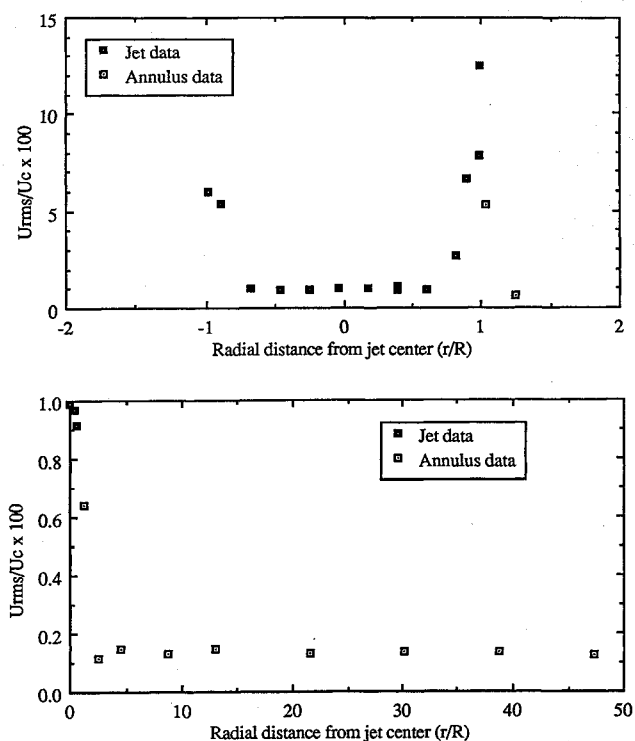


Fig. 2 Velocity fluctuation profile using argon jet at condition B,  $x/d = 0.4$ .

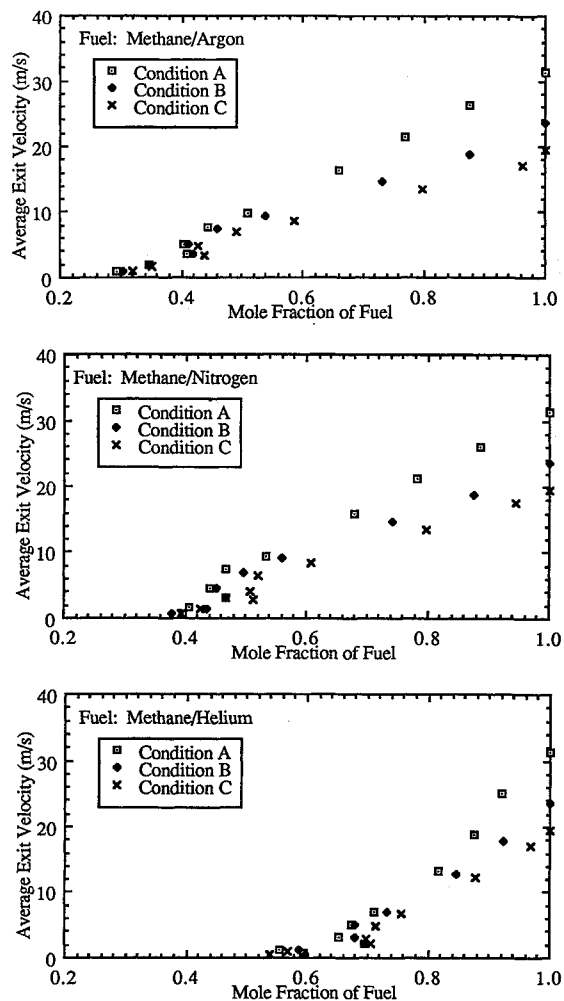


Fig. 3 Annular air velocity effects on methane jet diffusion flame lift-off.

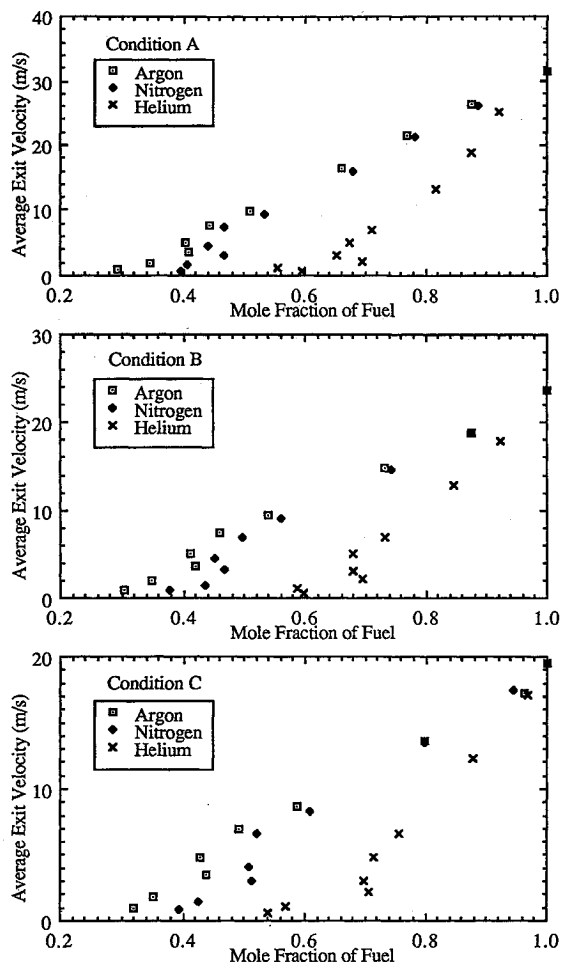


Fig. 4 Diluent effects on methane jet diffusion flame liftoff.

= 0.6. Similar observations are made for conditions B and C. The respective values at condition B are 23.7 m/s at  $X_F = 1$ , 0.99 m/s at  $X_{Ar} = 0.3$ , 0.82 m/s at  $X_{N_2} = 0.38$ , and 0.59 m/s at  $X_{He} = 0.6$ . Those values at condition C are, respectively, 19.5 m/s at  $X_F = 1$ , 0.97 m/s at  $X_{Ar} = 0.32$ , 0.79 m/s at  $X_{N_2} = 0.39$ , and 0.57 m/s at  $X_{He} = 0.54$ .

The data of Fig. 3 are replotted to show the diluent effects by Fig. 4. It appears that helium has the most destabilizing effects, i.e., the helium diluted jet has the lowest liftoff velocity when the same mole fraction of dilution is considered. Argon dilution shows the least destabilizing effect among the three inert gases examined. This observation is in contrast to previous work investigating the same diluents in a hydrogen

jet diffusion flame at liftoff.<sup>4,5</sup> The hydrogen jet diffusion flame liftoff was investigated using normal annular air<sup>4</sup> or substitute air.<sup>5</sup> The substitute air replaced nitrogen with argon or helium in composition. These two studies revealed that helium was the most stable, and argon the least stable among the three inert gases examined. Normal and substitute air had little, if any, influence on the liftoff characteristics of hydrogen jet diffusion flames.<sup>4,5</sup> The opposite trends in the liftoff velocities of hydrogen and methane jet diffusion flames using the same inerts are surprising. One additional note is that the laminar burning velocity has been recognized to be an important parameter in the liftoff of jet diffusion flames. The laminar burning velocity, however, increases with the substitute of helium in air for both hydrogen<sup>4</sup> and methane<sup>9</sup> flames.

### Acknowledgments

This work was supported by the Air Force Aerospace Sciences Directorate, by an Air Force Doctoral Fellowship to JPS (Contract F49620-86-C-0127/SB5861-0436), and by the National Science Foundation through an Engineering Research Equipment Grant (CTS-8922003). Special thanks are extended to L. P. Goss, B. Sarka, and C. Obringer.

### References

- <sup>1</sup>Seaba, J. P., "Burner Stabilities of Jet Diffusion Flames," Ph.D. Dissertation, Dept. of Mechanical Engineering, Univ. of Iowa, Iowa City, IA, 1990.
- <sup>2</sup>Wohl, K., Knapp, N., and Gazley, C., *Third Symposium (International) on Combustion, Flame, and Explosion Phenomena*, William and Wilkens Co., Madison, WI, Sept. 1948, p. 3.
- <sup>3</sup>Vanquickenborne, L., and Van Tiggelen, A., "The Stabilization Mechanism of Lifted Diffusion Flames," *Combustion and Flame*, Vol. 10, No. 1, 1966, pp. 59–69.
- <sup>4</sup>Takahashi, F., Mizomoto, M., and Ikai, S., "Stability of Hydrogen Jet Diffusion Flames," *Hydrogen Energy Progress*, Proceedings of the 3rd World Hydrogen Energy Conference, Tokyo, Japan, June 1980, pp. 1156–1176.
- <sup>5</sup>Takahashi, F., Mizomoto, M., Ikai, S., and Futaki, N., "Lifting Mechanism of Free Jet Diffusion Flames," *Twentieth Symposium (International) on Combustion*, The Combustion Inst., Ann Arbor, MI, Aug. 1984, pp. 295–302.
- <sup>6</sup>Takahashi, F., Mizomoto, M., Ikai, S., and Tsuruyama, K., "Stability Limits of Hydrogen/Air Coflow Jet Diffusion Flames," *AIAA Paper* 90-0034, Jan. 1990.
- <sup>7</sup>Takahashi, F., and Schmoll, W., "Lifting Criteria of Jet Diffusion Flames," *Twenty-Third Symposium (International) on Combustion*, The Combustion Inst., Orléans, France, July 1990, pp. 677–683.
- <sup>8</sup>Chen, L.-D., and Roquemore, W. M., "Visualization of Jet Flames," *Combustion and Flame*, Vol. 66, No. 1, 1986, pp. 81–86.
- <sup>9</sup>Clingman, W., Brokaw, R., and Pease, R., "Burning Velocities of Methane with Nitrogen-Oxygen, Argon-Oxygen, and Helium-Oxygen Mixtures," *Fourth Symposium (International) on Combustion*, William and Wilkens Co., Boston, MA, Sept. 1952, pp. 310–313.